

	Aerotrain Atmospheric	Aerotrain Vacuum tube	Hyperloop Alpha Atmospheric	Hyperloop Alpha Vacuum tube	
Gas Constant	8314 R	8314	8314 R	8314	
Gas Mol Wt (air)	28.97 M	28.97	28.97 M	28.97	
Heat cap ratio (air)	1.4 k	1.4	1.4 k	1.4	
Weight of train Kg	12,000	12,000	12,000	12,000	
Bum and curve g-loading	1.50	1.50	3.00	3.00	
Max loading Kg	18,000	18,000	36,000	36,000	
Length of each cushions m	20	20	1.50	1.50	
Width of each cushion m	1	1	0.90	0.90	
Number of cushions	4	4	28.00	28	
Gap height mm	25	25	1.00	1.00	1 mm slot is extremely small
Total area m2	80	80	37.80	37.80	
% of cushion used for lifting	60%	60%	90%	90%	
Pressure cushion relative Pa	3,679	3,679	10,381	10,381	Pressure above ambient
Total gap length m	168	168	134.4	134.4	
Total gap area m2	4.2	4.2	0.1344	0.1344	
Ambient pressure atmos Pa	101,000 P2	100	101,000 P2	100	
Pressure of cushion Pa	104,679 P1	3,779	111,381 P1	10,481	
Air temp K	298 T	298	400 T	400	
Air density Kg/m2	1.224 Dn	0.044	0.970 Dn	0.091	
Speed of sound m/s	346 Ss	346	401 Ss	401	Alpha higher due hig temp
Orifice flow coeff	0.8	0.8	0.8	0.8	
Pressure increase/input	0.04	36.79	0.10	103.81	
Is it Choked? Y or N	N	Y	N	Y	Choked above press ratio 2
Incompressible velocity m/s	62.02	326.45	117.03	381.49	Only valid at low speeds

Slot exit velocity m/s	62.02 C	346.02	117.03	400.89	Mach 1
Slot vol flow m3/s	260.50	1,453.29	15.73	53.88	
Slot mass flow kg/s	318.86	64.21	15.26	4.92	

Mass flow rate Kg/s	318.856	64.213	15.260	4.919
Inlet pressure Pa	101,000	100	101,000	100
Inlet temp K	292.0	292.0	292.0	292.0
Discharge pressure Pa	104,679	3,779	111,381	10,481
Polytropic Efficiency	0.810	0.810	0.810	0.810
Overall compression ratio	1.04	37.79	1.10	104.81
Polytropic exponent	1.54	1.54	1.54	1.54
density at inlet	1.20525	0.00119	1.20525	0.00119
Average compressibility=1	1.0000	1.0000	1.0000	1.0000
Discharge temp K	296	1,051	302	1,507
Discharge temp C	23	778	29	1,234
Polytropic head	3017	617850	8342	988343
Polytropic gas power W	1,187,639	48,980,245	157,159	6,002,415
Compressor power kW	1,188	48,980	157	6,002

Power ratio	41.2	Power ratio	38.2
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Multiplier of power required in vacuum vs atmosperic

Alpha exit speed	
Mass flow kg/s	0.2
Vol flow m3/s	2.19
Gap area m2	0.1344
Exit speed m/s	16.30

This is Alpha's flow

The purpose of this sheet is to show how much the compressor power must increase when an air cushion vehicle runs in a vacuum tube with 1/1000th of an atmosphere.

The first column is matching the achieved performance of the French Aerotrain  
The cushion sizes are estimated, then the gap width is adjusted to get about 1,100 kW for the lift compressors  
The gap is 90mm, this is required for stability and protection against skirt rubbing  
The big gap allows the airflow, and cushion pressure to respond to ride height  
The flow velocity is calculated using an incompressible orifice flow formula

The second column looks at the same configuration, but running in a partial vacuum of 100 Pa  
The compressor power increases by a factor of 43:1 to 46 mW

The third column is Hyperloop Alpha  
The gap height of 1mm is used  
When running in the vacuum tube, the power increase is a factor of 38:1, to 6 mW

All the compressor calcs were adapted from  
[www.elsevierdirect.com](http://www.elsevierdirect.com) 06\_Compressors.xls Power-Centrifugal.  
This compressor calculator matches Hyperloop Alpha's compressors. Polytropic efficiency may need to be adjusted.

Incompressible flow thru orifice, only applies for low pressure ratios  
 $Q = C A \sqrt{2(P_1 - P_2) / \rho}$   
but  $V = Q / A$   
 $V = C \sqrt{2(P_1 - P_2) / \rho}$

Choked flow theory, states that for pressure ratios above about 2, the flow velocity will be Mach 1.  
The Kantrowitz limit says that once Mach 1 is achieved, the speed will not increase regardless of lowering the outlet pressure.



The French Aerotrain achieved a remarkable 430 km/h in 1974





The British Hovertrain had lift pads under the skirts



There are several interesting ground-effect projects. They would not work in a vacuum